IN THE CLAIMS:

Please enter the following amended claims as follows:

- 1. (currently amended) A fuel injector nozzle for a gas turbine, the nozzle comprising an air supply presented to a fuel distribution arrangement passage centrally located by a wall having a selected cross section and a fuel supply passage surrounding said air supply passage and having a cross section greater than said cross section of said air supply passage so that, in operation, the air supply from the air supply passage will directly impinge on at least a portion of said nozzle, said fuel supply passage and said nozzle will have air flow portions of lower pressure outside of the portions of direct impingement and where the nozzle includes fuel distribution structures which are asymmetrically distributed about the nozzle to differentially present fuel to the whereby fuel presented to the air flow is mixed for subsequent combustion in use, the fuel being presented by fuel distribution structures in the nozzle wherein the fuel distribution structures are asymmetrically distributed about the nozzle whereby fuel is differentially presented to an air flow passing through the nozzle in use and dependent upon on localised localized air flow pressure.
- 2. (cancelled).
- 3. (currently amended) A nozzle as claimed in claim 1 wherein the fuel distribution structures are configured such that less fuel is presented [[at]] to the portions of the air flow of lower flow pressure typically outside of the direct impingement cross-section of the air flow than to the portions of air flow of higher pressure.
- 4. (cancelled).
- 5. (currently amended) A structure as claimed in claim [[4]] 1 wherein the fuel distribution structure comprises a plurality of grooves.
- 6. (currently amended) A structure as claimed in claim [[4]] 1 wherein the fuel distribution structure comprises a number of passageways.

- 7. (withdrawn and currently amended) A fuel distribution structure as claimed in claim [[4]] 1 wherein the fuel distribution structure comprises a number of apertures to appropriately present fuel to the air flow.
- 8. (withdrawn) A fuel distribution structure as claimed in claim 7 wherein the fuel distribution structure comprises a number of cross-section aperture portions, asymmetrically distributed about the fuel distribution structure.
- 9. (withdrawn) A fuel distribution structure as claimed in claim 7 wherein the fuel distribution structure comprises a number of variably different cross-section apertures evenly distributed about the fuel distribution structure.

 10. (cancelled).
- 11. (withdrawn and currently amended) A fuel distribution structure as claimed in claim [[10]] 1 wherein the fuel distribution structure is angled relative to the direction of air flow.
- 12. (currently amended) A fuel distribution structure as claimed in claim [[11]] 1 wherein the fuel distribution structure is angled relative to the direction of air flow.
- 13. (withdrawn) A fuel distribution structure comprising a number of elements having a height in the range 0.25 1.00mm, a width in the range 0.25 1.00mm and with a pitch between respective elements on the order of 3 20°.
- 14. (withdrawn) A fuel distribution structure incorporating cross-sectional portions as claimed in claim 8.
- 15. (currently amended) A turbine engine incorporating a fuel distribution structure as claimed in claim [[4]] 1.